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THE ROSCOE MANUAL

Volume 1-2: A Simplified ROSCOE Input Scheme

12

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20. (Continued)

The ROSCOE documentation consists of a number of volumes, including user manuals (Volumes 1 through 3), systems code descriptions (Volumes 4, 20, and 21-1), code validation documents (Volumes 6 and 23), and phenomenology code descriptions (all others). This document has been written as an extension to the user manuals. It describes a simplified input scheme for running a subset of ROSCOE problems. It is intended for the user who only occasionally runs the code or would like to run a small problem.

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1 INTRODUCTION

In the last few years, ROSCOE (Radar and Optical Systems Code with Nuclear Effects) has been expanded to include simulations of satellite communications and optical surveillance systems in a nuclear environment. This expansion has led to considerably more complexity in the input requirements.

While the ROSCOE input scheme was devised to handle these problems (with no additional coding) and to allow the user complete flexibility in structuring scenarios with multiple sensors, objects, and bursts, it takes some time to learn how to use the system. For the user who only occasionally runs the code, or would like to run a small problem, a new input scheme has been built for running a subset of ROSCOE problems with a simple set of inputs.

The next section describes this new input scheme. Example input sets are shown for several different types of problems and the program outputs are briefly discussed. Section 3 describes how to access the new scheme, for both batch and interactive jobs. Finally, to make this paper useful as a reference guide, tables which describe the input options have been placed in Appendix A.

2 DESCRIPTION

The new ROSCOE input scheme consists of a data deck with a pre-selected set of input options, and a data preprocessor program which inserts user-specified values for the options into the data deck. The scheme, in general, does not sacrifice any of ROSCOE's input versatility, since a new data deck with a different set of options can be generated without writing new code.

2.1 LIMITATIONS

With the new scheme, as currently set up, the user can run nuclear burst phenomenology problems alone, or nuclear effects on radar surveillance and tracking of ballistic missiles, satellite communication, or optical surveillance and tracking, subject to these constraints:

- Up to five bursts are allowed, at altitudes up to 400 km -- positions, times, and burst properties are input.
- Only one radar can be simulated in a run -- radar characteristics and location are input.
- Only one object trajectory can be simulated in a run (although multiple objects can be spaced in time on the trajectory)--launch and impact points, impact time, and reentry angle are input.
- Only one satellite communication system can be simulated in a run (consisting of one ground transmitter, one ground receiver, and one set of satellite-borne equipment which receives and transmits) -- transmitter and receiver characteristics and locations are input.
- Only one optical sensor can be simulated in a run -- sensor characteristics and location are input.
- Run times can be no more than 900 seconds after the last burst.

2.2 INPUT VARIABLES

Input variables in the new scheme are of five types:

- General Inputs. Variables related to a reference location or time.
- Physics Inputs. Variables required to simulate a burst and print physics outputs.
- Radar Inputs. Variables required to simulate radar surveillance or tracking performance.
- Satcom Inputs. Variables required to simulate a satellite communication problem.
- Optics Inputs. Variables required to simulate optical sensor surveillance or tracking performance.

Table A.1 is a directory of input variables, divided into the five types described above with notes to indicate the options available. For each variable, the table gives its name, the number of values to be supplied (more than one if the variable is a vector), a definition of the variable including default units of measure, the default values that will be assumed if you do not input the variable, and whether a unit name is allowed for the variable. (Table A.2 shows the allowable unit names.) It is important to note the default units given. If you input values without unit names (for those variables allowing unit names), the default units are assumed. Note that the default values listed in Table A.2 are given in their customary units, which are not always the same as the internal default units.

To run a case, follow the instructions given in Table A.1, and input those variables you wish to change in the form: variable = value unit, variable = value unit, etc. End the input string with the command RUN following the last variable input. For vectors, the format may be: vector = value unit, value unit, etc., or vector(index) = value unit, value unit, etc. In the first case, the values are assigned to vector(1),

vector(2), etc.; in the second case, values are assigned to vector(index), vector(index + 1), etc. This free format is essentially compatible with the Fortran NAMELIST input scheme.

Note that positions can be specified by geographical coordinates (GEOGR), or by Cartesian (LOCXYZ) or range-azimuth-elevation (RADAR) coordinates relative to a reference location. The order of entry, orientation, and units for these specifications are given in Table A.3 and Fig. A.1.

2.3 EXAMPLE INPUT SETS

2.3.1 Physics Problem

To run a simple physics problem consisting of a single burst with the default characteristics and these assumptions:

- Burst time = 0 s
- Yield = 10 kT
- Altitude = 40 km
- Output every 20 s until 120 s after burst

input:

```
TSTØP = 120, ØTIME = 0, ØTINT = 20, BTIME1 = 0,  
BPØS1(3) = 40, YIELD1 = 10 KT, RUN
```

2.3.2 Radar Problem

To run a radar surveillance problem, where:

- There is a single burst with the above properties.
- The radar is at the center of a local Cartesian coordinate system (directly under the burst).
- The radar is of the type described by the default parameters.
- The object being viewed has a -30° reentry angle and is aimed at the radar.

- The object is at 100 km altitude at time = 0 when the burst occurs.
- Radar measurements are made once every second for 20 s.

input:

TSTØP = 20, BTIME1 = 0, BPØS1(3) = 40, YIELD1 = 10 KT,
 ØBTAG = ØBJECT-1, ØBTIM = 0, ØBPØS(2) = 173,
 100 KM, ØBVEL(3) = -30, RADAR = REFER, RUN

2.3.3 Satcom Problem

To run a satellite communication problem, where:

- The ground transmitter and receiver are together, directly beneath a satellite at synchronous altitude (the default condition)
- The default link inputs are assumed
- The default nuclear burst (1 MT at 200 km altitude) occurs 10 s after the first communication
- The burst is displaced 200 km horizontally from the line of sight
- Communication calculations are made every 20 s, from 0 s to 100 s

input:

TSTØP = 100, BTIME1 = 10, BPØS1(2) = 200, CTIME = 0,
 CTINT = 20, RUN

2.3.4 Optics Problem

To run an optical sensor surveillance problem, where:

- There is a single burst of 10 kt at 40 km altitude
- The sensor is at synchronous altitude
- The sensor is pointed at the burst

- The sensor is of the type described by the default parameters
- Sensor calculations are made at only one time (0 s)

input:

TSTØP = 1, BTIME1 = 0, BPØS1(3) = 40, YIELD1 = 10 KT,
 ØBTAG = REF-ØBJECT, ØTYPE = SURVEILLANCE, ØLØØK = 0,
 REFPT(1) = 40, ØPTICS = REFER, RUN

2.4 OUTPUTS

The outputs produced by the ROSCOE code using the new input scheme are described in this section. Two types of outputs may be produced:

(1) printer plots, and (2) tabular outputs.

2.4.1 Printer Plots

When a high-altitude (>90 km) burst is simulated, the code produces a series of printer plots at times specified by the ØTIME, ØTINT input variables. The plots consist of a picture of the fireball and beta tube region and contour plots of mass density, electron density, and striation fraction in the high-altitude grid.

The contour plots of mass density and electron density represent vertical cross sections through the burst point in the (magnetic) north-south direction, viewed looking eastward. The contour plots of the striation fraction are cross sections normal to the earth's magnetic field, viewed looking down the field lines.

The plots are produced as they are computed internally, and thus will appear before the tabular output described below.

In addition, contour plots of the relative radiance at the focal plane of the sensor can be generated when the optics code is used. These plots are generated when the optics calculation type, OCA1, is set to FOV.

2.4.2 Tabular Outputs

There are seven phenomenology lists, five radar lists, two satellite-communication lists, and three optics lists that may be output at the conclusion of the run, depending on the type of simulation performed.

The phenomenology lists include: burst parameters, common fireball parameters (fireball set 1), two additional low-altitude fireball parameter lists (fireball set 2 and fireball set 3), additional high-altitude fireball parameters (fireball set 4), contained debris region parameters, and beta tube parameters.

The radar lists include: trajectory output, track measurement errors, track filter output, and two lists of propagation errors.

The satcom lists include: propagation and probability-of-error data, and satellite position coordinates with respect to the ground-terminal positions.

The optics output lists include: angle and signal-strength measurements for an optical tracking sensor application, the radiance along each path treated within the field-of-view, and the data stream output produced by a scanning sensor.

Table 1 shows a small sample of each type of output. Some of the column headings are self-explanatory, while others require additional comment.

TABLE 1

ROSCOE TABULAR OUTPUTS

PHENOMENOLOGY OUTPUTS

BURST PARAMETERS

TIME OF OUTPUT SEC	TOTAL ENERGY (ERGS)	FISSILE ENERGY (ERGS)	BURST ALTITUDE KM	BURST DENSITY (GM/CC)	SCALE HEIGHT KM	BURST PT. TEMP (DEG K)	INITIAL RADIUS KM	TIME TO REACH 3000K	TIME TO REACH 2000K
1030.000	.4193E+23	.2092E+23	250.000	.5045E+13	37.579	39037.792	212.560	0.000	0.000
1050.000	.4193E+23	.2092E+23	250.000	.5045E+12	67.922	13085.556	98.567	0.000	0.000

NOTE: Columns 9 and 10: The outputs "time to reach 3000 K and 2000 K" are used only for low-altitude (<90 km) fireball chemistry calculations.

FIREBALL SET-1

TIME OF OUTPUT SEC	FIREBALL INDEX NUMBER	HORIZONTAL RADIUS KM	VERTICAL RADIUS KM	CENTER ALTITUDE KM	RISE RATE KM	EXPANSION RATE KM	FIREBALL DENSITY (GM/CC)	FIREBALL TEMP (DEG-K)	TIME SINCE BURST SEC
1030.000	1	280.583	269.073	421.580	1.671	0.000	.3021E-12	11213.966	10.000
1050.000	1	280.583	626.172	475.551	1.475	0.000	.3688E-13	8724.484	50.000

FIREBALL SET-2

TIME OF OUTPUT SEC	FIREBALL INDEX NUMBER	MINIMUM ALTITUDE KM	MAXIMUM ALTITUDE KM	TILT FROM VERTICAL DEG	AXIS ROTATION DEG	MCW VORTEX RADIUS KM	VRT VORTEX RADIUS KM	VORTEX VOLUME (CM ³)	CHARACT. TYPE SEC
1030.000	1	166.278	690.601	.713	0.000	280.533	269.073	.0055E+23	0.000
1050.000	1	165.783	901.723	5.500	0.000	280.533	426.172	.1290E+24	0.000

NOTE: Column 6: Axis rotation is measured +CCW from North. Column 10: The characteristic time is the approximate time this fireball has merged with another (used only for low-altitude fireballs).

Table 1 (Continued)

FIREBALL SET-3										
TIME OF OUTPUT SEC	FIREBALL INDEX NUMBER	X- COORDINATE (CM)	Y- COORDINATE (CM)	Z- COORDINATE (CM)	OVAL OF CASSINI PARAMETER	OVAL ARM RADIUS KM	VERTICAL INDEX (Z=010.)	FIREBALL KIND	MERGE TO INDEX	
95.000	1	.1150E+09	.0001E+09	.0210E+09	.050	0.000	497.913	1	0	
96.000	1	.1150E+09	.0001E+09	.0210E+09	.051	0.000	498.233	1	0	

NOTE: Column 6: The Oval of Cassini parameter describes the shape of a low-altitude fireball. A value of 1.0 or greater means the fireball has formed a torus. Columns 9 and 10: The fireball kind can take values from 1 to 5, where: 1 = spheroid, 2 = skewed spheroid, 3 = torus, 4 = inactive radiation-merged fireball, 5 = inactive hydromerged fireball.

FIREBALL SET-4										
TIME OF OUTPUT SEC	FIREBALL INDEX NUMBER	X- COORDINATE (CM)	Y- COORDINATE (CM)	Z- COORDINATE (CM)	GRID CELL INDEX (X=010.)	GRID CELL INDEX (Y=010.)	GRID CELL INDEX (Z=010.)	FIREBALL REL. PCS. IN CELL	FIREBALL KIND	
1030.000	1	.0001E+09	.0001E+09	.0001E+09	3	3	3	.094	2	
1050.000	1	.0520E+09	.0527E+09	.0507E+09	3	3	3	.094	2	

NOTE: Columns 6 to 8: The grid cell indices refer to the grid cell in which the fireball center is located.

DEBRIS PARAMETERS										
TIME OF OUTPUT SEC	FIREBALL INDEX NUMBER	DEBRIS INDEX NUMBER	TOTAL ENERGY (E=GS)	DEBRIS ALTITUDE KM	HORIZONTAL RADIUS KM	VERTICAL RADIUS KM	DEBRIS DISTRIBUTION PARAMETER	EQUIVALENT SPM, RAD. IN	DEBRIS VOLUME (C-3)	
95.000	1	1	.0100E+20	8.820	.053	.052	8.000	.052	.001E+12	
96.000	1	1	.0100E+20	8.820	.050	.050	8.000	.057	.210E+13	

NOTE: Column 8: The debris distribution parameter describes the rate of fall-off of the beta source strength from the tube boundary (see RANC IV).

Table 1 (Continued)

BETA TUBE PARAMETERS									
TIME OF OUTPUT SEC	FLYERALL INDEX NUMBER	RETATURE SHAPE	INITIAL DIP ANGLE DEG	ATK ANGLE FROM HORIZ DEG	KINK-BURST DISTANCE KM	N-S RADIUS AT 85KM KM	E-W RADIUS AT 85KM KM	N-S RADIUS AT 60KM KM	E-W RADIUS AT 60KM KM
1630.000	1	KINK	76.106	76.674	40.259	203.819	207.111	202.547	206.309
1650.000	1	KINK	76.106	77.821	88.328	201.169	205.439	199.915	204.644

NOTE: Column 3: The beta tube shape is either "STRAIGHT" or "KINK". Column 6: The kink-burst distance is the distance from the sub-burst point at 85 km to the center of the beta tube at 85 km.

RADAR OUTPUTS

TRAJECTORY OUTPUT

TYPE OF EVENT	TIME OF OUTPUT SEC	POSITION ALTITUDE M	DATA FOR RANGE M	OBJECT AT AZIMUTH DEG	SPECIFIED ELEVATION DEG	TIME VELOCITY M	SIGNAL TO NOISE (DB)	NUMBER OF TARGETS
SEARCH	1599.097	3296670.933	3296670.933	81.224	2.722	6226.094	19.879	1
VERIFY	1599.597	3296670.933	3296670.933	81.223	2.725	6226.259	20.575	1

NOTE: Column 1: The event type is either "SEARCH", "VERIFY", "TRACK IN" (for track initiation), or "TRACK". Columns 3 to 7: The position and velocity data given here are the actual values. Column 9: The number of targets can be zero if the target has been lost, one if a single target has been located, or more than one if multipath effects occur.

TRACK MEASUREMENT ERRORS

TIME OF OUTPUT SEC	PREDICTED RANGE M	PREDICTED AZIMUTH DEG	PREDICTED ELEVATION DEG	MEASURED RANGE M	MEASURED AZIMUTH DEG	MEASURED ELEVATION DEG	MEASUREMENT ELEVATION DEG
1599.097	3296670.933	81.224	2.722	3296670.939	81.182	2.604	0.000
1599.597	3296670.933	81.223	2.725	3296671.211	81.142	2.604	0.000

NOTE: Columns 2 to 4 and 5 to 7: The predicted position is either equivalent to the actual position for search pulses or is the position predicted by the track filter once track has been initialized. The measured coordinates are those generated during the current look and include all refraction and radar measurement errors.

Table 1 (Continued)

TRACKING ERRORS												
TIME OF OUTPUT SEC	ERRORS IN ALONG V M	POSITIONS PERP TO V M	ERRORS IN ALONG V M	VELOCITIES PERP TO V M	CLUTTER LOSS	DISPERSIVE LOSS	APPARENT RANGE M	TARGET AZIMUTH DEG	POSITION ELEVATION DEG			
1416.497	512.146	1504.945	-57.792	1718.544	6084.306	-2792.826	3193576.724	80.954	3.250			
1417.497	112.660	913.467	1943.290	451.061	2202.409	384.810	3187482.417	80.938	3.280			

NOTE: Columns 2 to 7 and 8 to 10: The errors in position and velocity are the difference between the filter prediction and actuals. The apparent target coordinates are the actual coordinates plus refraction and multi-path errors before radar measurement errors have been added.

PROPAGATION OUTPUT--I									
TIME OF OUTPUT SEC	ABSORPTION FROM ALL SOURCES	THRESHOLD ANSORPTION	NOISE TEMP.	NOISE POWER	CLUTTER TO-NOISE RATIO (DB)	DISPERSIVE LOSS	FARADAY ROTATION LOSS	FAILURE MODE	
1509.497	0.000	7.206	0.000	.2445E-09	0.000	1.000	1.000	NO FAILURE	
1500.597	0.000	7.563	0.000	.2445E-09	0.000	1.000	1.000	NO FAILURE	

NOTE: Column 9: The failure mode flag can have the following messages:

NO FAILURE	S/N received is above threshold
RANGE	The radar is range (power) limited for this target
ABSORPTION	The absorption due to all sources has reduced the S/N below threshold
ABS+NOISE	The combination of absorption and fireball noise has reduced the S/N below threshold
TOTAL	The combination of absorption, noise, dispersion, and Faraday rotation has dropped the S/N below threshold
LOW SIGNAL	The combination of the above effects and refraction or clutter has dropped the S/N below threshold
NO TARGET	There are no targets within the range gate and 3 dB beamwidth

Table 1 (Continued)

PROPAGATION OUTPUT-2									
TIME OF OUTPUT SEC	RANGE M	BIAS RANGE M	REFRACTION AZIMUTH DEG	ERRORS ELEVATION DEG	RANDOM RANGE M	REFRACTION AZIMUTH DEG	ERRORS ELEVATION DEG		
1500.007	0.000	0.000	0.000	0.000	0.000	0.000	0.000	0.000	0.000
1502.507	0.000	0.000	0.000	0.000	0.000	0.000	0.000	0.000	0.000
SATCOM OUTPUTS									
COMMUNICATIONS OUTPUT -1									
TYPE OF OUTPUT	TIME OF OUTPUT SEC	UPLINK LOSS FACTOR	UPLINK SCINY	DOWNLINK LOSS FACTOR	DOWNLINK SCINY	PROR. OF ERROR SATELLITE	PROR. OF ERROR GROUND	PROR. OF ERROR	PROR. OF ERROR
CCW-DECEVO	1612.000	1.001	0.	1.605	0.	0.	0.	0.	0.
CCW-DECEVO	1622.000	22.903	9526.6	60.143	10230.	0.25805E-03	.12247E-01	.12500E-01	.12500E-01

NOTE: Columns 3 to 6: The uplink and downlink loss factors are the losses due to absorption from all sources (dimensionless). The uplink and downlink scintillation values refer to the standard deviation in phase due to scintillation effects in radians.

COMMUNICATIONS OUTPUT -2									
TIME OF OUTPUT SEC	SATELLITE RANGE KM	COORD. RT AZIMUTH DEG	TRANSMIT ELEVATION DEG	SATELLITE RANGE KM	COORD. RT AZIMUTH DEG	RECEIVED ELEVATION DEG			
1612.000	1306.105	-86.892	74.592	1306.105	-86.892	74.592			
1622.000	1291.663	-86.922	77.716	1291.663	-86.922	77.716			

Table 1 (Continued)

OPTICS OUTPUT

OPTICAL MEASUREMENTS									
TYPE OF OUTPUT	CENTRAL WAVELENGTH (MICRONS)	ACTUAL AZIMUTH (DEGREES)	ESTIMATED ELEVATION (DEGREES)	MEASURED AZIMUTH (DEGREES)	ESTIMATED ELEVATION (DEGREES)	SCATTERING COEFFICIENT (W/CM ²)	IRRADIANCE AT SENSOR (W/CM ²)	SCATTERING COEFFICIENT (W/CM ²)	SIGNAL-TO-NOISE RATIO (DB)
0.000	0.1957	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0

NOTE: Actual, measured, and estimated coordinates are measured in angular units relative to the sensor boresight.

INTEGRATED PATH DATA

TYPE OF OUTPUT	CENTRAL WAVELENGTH (MICRONS)	ACTUAL AZIMUTH (DEGREES)	ESTIMATED ELEVATION (DEGREES)	INTEGRATED RADIANCE (W/CM ²)	INTEGRATED RADIANCE (W/CM ²)	INTEGRATED RADIANCE (W/CM ²)
0.000	0.1957	0.0	0.0	0.0	0.0	0.0
0.000	0.1957	0.0	0.0	0.0	0.0	0.0
0.000	0.1957	0.0	0.0	0.0	0.0	0.0

NOTE: The radiance in Column 5 is the integrated radiance along the path (described by the azimuth and elevation off-boresight) due to all emission and scattering sources. The integrated radiance in Column 6 is just radiance integrated over all band intervals and the signal due to structure (Column 7) is the deviation in the integrated radiance due to striated (or structured) regions along the path.

OPTICAL SAMPLES

TYPE OF OUTPUT	CENTRAL WAVELENGTH (MICRONS)	ACTUAL AZIMUTH (DEGREES)	ESTIMATED ELEVATION (DEGREES)	SCATTERING COEFFICIENT (W/CM ²)	IRRADIANCE AT SENSOR (W/CM ²)	SCATTERING COEFFICIENT (W/CM ²)	SIGNAL-TO-NOISE RATIO (DB)
0.00000	0.1957	0.0	0.0	0.0	0.0	0.0	0.0
0.00010	0.1957	0.0	0.0	0.0	0.0	0.0	0.0
0.00020	0.1957	0.0	0.0	0.0	0.0	0.0	0.0

NOTE: The last four columns show the scanned signal output (irradiance at the detector), the normalized signal output (the irradiance normalized to the sensor NEFD), the final signal output (after all other processing such as differencing has been completed), and the target detection flag which signifies whether the final signal exceeds a preset threshold designating the point a "TARGET" versus a "BKGRND" point.

3 ACCESSING THE INPUT SCHEME

3.1 BATCH JOBS

To access and use the new input scheme in the batch mode (i.e., by submitting a card input deck over the counter or through a remote terminal), use a deck setup such as that shown in Table A.4.

Note that an optional card may precede the data cards, directing the input program to print each default card changed, followed by the new card which replaces it.

3.2 INTERACTIVE JOBS

To access and use the new input scheme using the time-share system follow these steps (also shown in Table A.6). (First, you must have a procedure permfile containing a small CYBER control language "PROC" and a set of control cards. A sample procedure permfile is shown in Table A.5)

- Step 1. Access your procedure file with the ATTACH statement.
- Step 2. Execute the ROSCOE time-share program by typing RØSCØTS.
- Step 3. Type your inputs, in response to the program's message "INPUTS?". The program then processes the inputs; that is, inserts them into the standard deck and checks for errors. If errors occur, the program prints them and asks you to input a revised list by again asking "INPUTS?". When no errors occur, terminate RØSCØTS by typing "RUN". The job file is then automatically placed in the input queue, and control returns to the INTERCOM system. You can check that your job has been accepted by typing a FIND, nnn command, where nnn is the first 1-5 characters of the job name (first parameter on your first control card).

APPENDIX A

USER REFERENCE TABLES

DIRRECTORY OF INPUT VARIABLES		TABLE A.1		02/04/80		10.56.10.		PAGE 1	
INPUT VARIABLE		A DIRECTORY OF INPUT VARIABLES				DEFAULT VALUES		UNIT-NAME ALLOWED	
VALUES		DESCRIPTION							

SIMPLIFIED MCSOCE INPUT LIST

A. GENERAL INPUTS

--THE DEFAULT VALUES ARE SET SO THAT THE CODE PROCESSES THE STOP EVENT FIRST AND THEN PROCESSES NO OUTPUT.
 --TO RUN A PHYSICS AND/OR RAFA, SATCOM, OR OPTICS PROBLEM CHANGE THE EVENT TIMES DESCRIBED BELOW TO OCCUR BEFORE THE STOP TIME.

|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|

DIRECTORY OF INPUT VARIABLES
Table A.1 (Continued)

INPUT VARIABLE	NC.	VALUES	DESCRIPTION	DEFAULT VALUES	UNIT-NAMF ALLOWED
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1. RUN CONTROL

--THE CCCE PROVIDES ONLY BURST PARAMETER OUTPUT BY
DEFAULT. TO GET TIME DEPENDENT FIREBALL AND CLOUD
PROPERTIES AT REGULAR INTERVALS, INPUT (UTIME) AND
(OTINT).
--FOR EXAMPLE--BY INPUTTING OTIME=0, SEC.OTIME=1, SEC
OTIME=30, SEC, THE CCCE WILL OUTPUT PHYSICS DATA
STARTING AT 1 SEC AFTER BURST AND PRINT OUTPUT EVERY 30
SEC THEREAFTER UNTIL THE STOP TIME IS REACHED.

OTIME	1	OTIME = PHYSICS OUTPUT TIME (DEFAULT UNIT IS SEC)	99999.	SEC	YES
OTINT	1	OTINT = PHYSICS OUTPUT DATA TIME INTERVAL (DEFAULT UNIT IS SEC)	30.	SEC	YES

2. BURST DATA

--UP TO FIVE BURSTS ARE ALLOWED.
--THE USER CAN CHOOSE TO INPUT BURST COORDINATES IN
GEOGRAPHICAL COORDINATES (GCOORD) OR COORDINATES
RELATIVE TO THE REFERENCE LOCATION IN SECTION A. ABOVE
(LOCKYZ OR MACAR).
--FOR EXAMPLE--THE USER CAN INPUT BPOS1=0.,50.,50.,LOCKYZ
AND BURST 1 IS SPECIFIED IN A CARTESIAN EAST-NORTH-UP
(XYZ) COORDINATE SYSTEM.

BTIME1	1	BTIME1 = BURST TIME FOR BURST 1 (DEFAULT UNIT IS SEC)	99999.	SEC	YES
BTIME2	1	BTIME2 = BURST TIME FOR BURST 2 (DEFAULT UNIT IS SEC)	99999.	SEC	YES
BTIME3	1	BTIME3 = BURST TIME FOR BURST 3 (DEFAULT UNIT IS SEC)	99999.	SEC	YES
BTIME4	1	BTIME4 = BURST TIME FOR BURST 4 (DEFAULT UNIT IS SEC)	99999.	SEC	YES
BTIME5	1	BTIME5 = BURST TIME FOR BURST 5 (DEFAULT UNIT IS SEC)	99999.	SEC	YES
BPOS1(1-4)	4	BPOS1(1-4) = BURST COORDINATES FOR BURST 1. COORDINATE(1)=ALTITUDE(KM), X-COORDINATE, OR RANGE(KM). COORDINATE(2)=FAST LONGITUDE(DEG), Y-COORDINATE, OR AZIMUTH CCN FROM EAST(DEG). COORDINATE(3)= NORTH LATITUDE(DEG), Z-COORDINATE, OR ELEVATION(DEG). COORDINATE(4)=CCOORD TYPE (GCOORD, LOCKYZ, OR MACAR).	0., 0., 200., LOCKYZ	NO NO NO NO	NO NO NO NO
BPOS2	4	BPOS2(1-4) = BURST COORDINATES FOR BURST 2 (SEE BPOS1 DESCRIPTION)	0., 0., 200., LOCKYZ	NO NO NO NO	NO NO NO NO
BPOS3	4	BPOS3(1-4) = BURST COORDINATES FOR BURST 3 (SEE BPOS1 DESCRIPTION)	0., 0., 200., LOCKYZ	NO NO NO NO	NO NO NO NO
BPOS4	4	BPOS4(1-4) = BURST COORDINATES FOR BURST 4 (SEE BPOS1 DESCRIPTION)	0., 0., 200., LOCKYZ	NO NO NO NO	NO NO NO NO
BPOS5	4	BPOS5(1-4) = BURST COORDINATES FOR BURST 5 (SEE BPOS1 DESCRIPTION)	0., 0., 200., LOCKYZ	NO NO NO NO	NO NO NO NO

DIRECTORY OF INPUT VARIABLES
 Table A.1 (Continued)
 INPUT
 AC.

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 DEFAULT UNIT-NAME
 VALUES ALLOWED

DESCRIPTION

0.
 200.
 LOCXYZ

3. NEAPCN DATA

--UP TO FIVE NEAPCN TYPES CAN BE ENTERED.
 --X-RAY TEMPERATURES ARE LIMITED TO THE THREE VALUES
 LISTED.

YIELD1	1	YIELD1 = YIELD OF BURST 1 (DEFAULT UNIT IS MT)	1.0	MT	YES
YIELD2	1	YIELD2 = YIELD OF BURST 2 (DEFAULT UNIT IS MT)	1.0	MT	YES
YIELD3	1	YIELD3 = YIELD OF BURST 3 (DEFAULT UNIT IS MT)	1.0	MT	YES
YIELD4	1	YIELD4 = YIELD OF BURST 4 (DEFAULT UNIT IS MT)	1.0	MT	YES
YIELD5	1	YIELD5 = YIELD OF BURST 5 (DEFAULT UNIT IS MT)	1.0	MT	YES
FFRAC1	1	FFRAC1 = FISSION FRACTION OF BURST 1	.10		NO
FFRAC2	1	FFRAC2 = FISSION FRACTION OF BURST 2	.10		NO
FFRAC3	1	FFRAC3 = FISSION FRACTION OF BURST 3	.10		NO
FFRAC4	1	FFRAC4 = FISSION FRACTION OF BURST 4	.10		NO
FFRAC5	1	FFRAC5 = FISSION FRACTION OF BURST 5	.10		NO
HFRAC1	1	HFRAC1 = HYDRO FRACTION OF BURST 1	.24		NO
HFRAC2	1	HFRAC2 = HYDRO FRACTION OF BURST 2	.24		NO
HFRAC3	1	HFRAC3 = HYDRO FRACTION OF BURST 3	.24		NO
HFRAC4	1	HFRAC4 = HYDRO FRACTION OF BURST 4	.24		NO
HFRAC5	1	HFRAC5 = HYDRO FRACTION OF BURST 5	.24		NO
NFRAC1	1	NFRAC1 = NEUTRON FRACTION OF BURST 1	.01		NO
NFRAC2	1	NFRAC2 = NEUTRON FRACTION OF BURST 2	.01		NO
NFRAC3	1	NFRAC3 = NEUTRON FRACTION OF BURST 3	.01		NO
NFRAC4	1	NFRAC4 = NEUTRON FRACTION OF BURST 4	.01		NO
NFRAC5	1	NFRAC5 = NEUTRON FRACTION OF BURST 5	.01		NO
XFRAC1	1	XFRAC1 = X-RAY FRACTION OF BURST 1	.75		NO
XFRAC2	1	XFRAC2 = X-RAY FRACTION OF BURST 2	.75		NO
XFRAC3	1	XFRAC3 = X-RAY FRACTION OF BURST 3	.75		NO
XFRAC4	1	XFRAC4 = X-RAY FRACTION OF BURST 4	.75		NO
XFRAC5	1	XFRAC5 = X-RAY FRACTION OF BURST 5	.75		NO
GFRAC1	1	GFRAC1 = GAMMA FRACTION OF BURST 1	.001		NO
GFRAC2	1	GFRAC2 = GAMMA FRACTION OF BURST 2	.001		NO
GFRAC3	1	GFRAC3 = GAMMA FRACTION OF BURST 3	.001		NO
GFRAC4	1	GFRAC4 = GAMMA FRACTION OF BURST 4	.001		NO
GFRAC5	1	GFRAC5 = GAMMA FRACTION OF BURST 5	.001		NO
MASS1	1	MASS1 = NEAPCN MASS OF BURST 1 (DEFAULT UNIT IS GM)	1.516	GM	YES
MASS2	1	MASS2 = NEAPCN MASS OF BURST 2 (DEFAULT UNIT IS GM)	1.516	GM	YES
MASS3	1	MASS3 = NEAPCN MASS OF BURST 3 (DEFAULT UNIT IS GM)	1.516	GM	YES
MASS4	1	MASS4 = NEAPCN MASS OF BURST 4 (DEFAULT UNIT IS GM)	1.516	GM	YES
MASS5	1	MASS5 = NEAPCN MASS OF BURST 5 (DEFAULT UNIT IS GM)	1.516	GM	YES
XTM1	1	XTM1 = X-RAY TEMPERATURE (KEV) OF BURST 1 -- (0.5,1.0, OR 2.0) (MUST BE TYPED LITERALLY -- E.G. 0.5 ,NOT .5)	1.0		NO

• 1771-1800 •

C. RAUAM CODE I.I.F.U.T.S

--TC RLA A RALAR PROBLEM THE USER MUST FIRST SET
 RALAR=KEEP

---MAGNAR FACILITIES CAN THEN BE SET UP IN TWO WAYS--
(1) AN OBJECT POSITION, VELOCITY AND TIME CAN BE INPUT
(SEE OBJECT DATA - START VECTOR INPUT BELOW) AND THE
FIRST MAGNAR LOCK WILL BE INITIATED AT THE OBJECT TIME
SPECIFIED, OR
(2) THE USER SPECIFIES AN OBJECT TRAJECTORY (SEE OBJECT
DATA - TRAJECTORY INPUT) AND THE FIRST MAGNAR LOCK IS
ESTABLISHED WHEN THE OBJECT ENTERS THE MAGNAR FOV.

1. RUN CONTROL

---REFERENCE IT MAJAN LOOKS ON CLOSED LCCP TRACK
IS SELECTED BY SETTING (MFLG),
---SUBSEQUENT LOOKS ARE CREATED INTERNALLY EVERY (DT) SEC.

```

1  RADAR = FLAG FOR INITIALIZING RADAR PROBLEM (SET RADAR=REFER FOR
      RADAR CALCULATIONS)
1  MFLAG = FLAG FOR CLOSED LOOP TRACK (0=TRACK, 1=NO TRACK/SEARCH ONLY)
1  CT = RADAR LOOK (ON TRACK) INTERVAL (DEFAULT UNIT IS SEC)
1  DT

```

2. KADAK UATA

---THE HAZAM LOCATION CAN BE INPUT IN GEOGRAPHICAL COORDINATES (UCCOR) OR RELATIVE TO THE REFERENCE LOCATION IN SECTION A. ABOVE.
---FOR EXAMPLE--- THE DEFAULT VALUES FOR RDPOS PLACE THE HAZAM AT THE ORIGIN OF THE CARTESIAN EAST-NORTH-UP

DIRECTORY OF INPUT VARIABLES				02/04/80 10.56.18.		PAGE 7	
Table A.1 (Continued)				DEFAULT VALUES		UNIT-NAME ALLOWED	
INPUT VARIABLE				DESCRIPTION			
VALUES							
				--SUBSEQUENT SATCOM CALCULATIONS ARE PERFORMED EVERY (CTINT) SEC.			
CTIME	1	CTIME = FIRST SAT-COM CALCULATION TIME (DEFAULT UNIT IS SEC)		99999.	SEC	YES	YES
CTINT	1	CTINT = TIME STEP FOR SAT-COM CALCULATIONS (DEFAULT UNIT IS SEC)		30.	SEC	YES	YES
				2. PROCESSING DATA			
				--FOR A MORE DETAILED DESCRIPTION OF THESE INPUTS SEE THE RESCUE MANUAL VOL. 20.			
CTYPE	1	CTYPE = SATCOM MODULATION TYPE (QPSK, CPFSK, OR FSK)		CPSK		NO	NO
REGEN	1	REGEN = FLAG FOR REGENERATION OF SIGNAL AT SATELLITE (YES OR NO)		YES		NO	NO
CCONT	1	CCONT = FLAG FOR CURRENT FSK MODULATION (YES OR NO)		NO		NO	NO
DETERM	1	DETERM = FLAG FOR FULLY DETERMINISTIC MODE CALCULATIONS (YES OR NO)		YES		NO	NO
ORDER	1	ORDER = ORDER OF PHASE LOCKED LOOP (IF FIRST OR SECOND)		FIRST		NO	NO
				3. PLATFORM DATA			
				--RELATIVE COORDINATES CAN BE USED HERE TO ALIGN THE COMMUNICATIONS LINKS AND BURST REGIONS.			
XPOS	4	XPOS(1-4) = TRANSMITTER POSITION (XPOS(1-3)=POSITION COORD, XPOS(4)=COORD TYPE--GEOD, LOCKY2, OR RADAR) (DISTANCES IN KM, ANGLES IN DEG)		0.		NO	NO
				0.		NO	NO
				LOCKY2		NO	NO
				0.		NO	NO
				LOCKY2		NO	NO
				0.		NO	NO
				LOCKY2		NO	NO
				0.		NO	NO
				LOCKY2		NO	NO
				0.		NO	NO
				LOCKY2		NO	NO
				0.		NO	NO
				LOCKY2		NO	NO
				0.		NO	NO
				LOCKY2		NO	NO
				0.		NO	NO
				LOCKY2		NO	NO
				0.		NO	NO
				LOCKY2		NO	NO
				0.		NO	NO
				LOCKY2		NO	NO
				0.		NO	NO
				LOCKY2		NO	NO
				0.		NO	NO
				LOCKY2		NO	NO
				0.		NO	NO
				LOCKY2		NO	NO
				0.		NO	NO
				LOCKY2		NO	NO
				0.		NO	NO
				LOCKY2		NO	NO
				0.		NO	NO
				LOCKY2		NO	NO
				0.		NO	NO
				LOCKY2		NO	NO
				0.		NO	NO
				LOCKY2		NO	NO
				0.		NO	NO
				LOCKY2		NO	NO
				0.		NO	NO
				LOCKY2		NO	NO
				0.		NO	NO
				LOCKY2		NO	NO
				0.		NO	NO
				LOCKY2		NO	NO
				0.		NO	NO
				LOCKY2		NO	NO
				0.		NO	NO
				LOCKY2		NO	NO
				0.		NO	NO
				LOCKY2		NO	NO
				0.		NO	NO
				LOCKY2		NO	NO
				0.		NO	NO
				LOCKY2		NO	NO
				0.		NO	NO
				LOCKY2		NO	NO
				0.		NO	NO
				LOCKY2		NO	NO
				0.		NO	NO
				LOCKY2		NO	NO
				0.		NO	NO
				LOCKY2		NO	NO
				0.		NO	NO
				LOCKY2		NO	NO
				0.		NO	NO
				LOCKY2		NO	NO
				0.		NO	NO
				LOCKY2		NO	NO
				0.		NO	NO
				LOCKY2		NO	NO
				0.		NO	NO
				LOCKY2		NO	NO
				0.		NO	NO
				LOCKY2		NO	NO
				0.		NO	NO
				LOCKY2		NO	NO
				0.		NO	NO
				LOCKY2		NO	NO
				0.		NO	NO
				LOCKY2		NO	NO
				0.		NO	NO
				LOCKY2		NO	NO
				0.		NO	NO
				LOCKY2		NO	NO
				0.		NO	NO
				LOCKY2		NO	NO
				0.		NO	NO
				LOCKY2		NO	NO
				0.		NO	NO
				LOCKY2		NO	NO
				0.		NO	NO
				LOCKY2		NO	NO
				0.		NO	NO
				LOCKY2		NO	NO
				0.		NO	NO
				LOCKY2		NO	NO
				0.		NO	NO
				LOCKY2		NO	NO
				0.		NO	NO
				LOCKY2		NO	NO
				0.		NO	NO
				LOCKY2		NO	NO
				0.		NO	NO
				LOCKY2		NO	NO
				0.		NO	NO
				LOCKY2		NO	NO
				0.		NO	NO
				LOCKY2		NO	NO
				0.		NO	NO
				LOCKY2		NO	NO
				0.		NO	NO
				LOCKY2		NO	NO
				0.		NO	NO

DIRECTORY OF INPUT VARIABLES				02/04/80		10.36.1e.		PAGE 8	
Table A.1 (Continued)						DEFAULT		UNIT-NAME	
INPUT	NO.	VALUES	DESCRIPTION			VALUES		ALLOC-ED	
VARIABLE									
BITP	2		BIT PERIOD (UPLINK, DOWNLINK) (DEFAULT UNIT IS SEC)			1.0E-8		SEC	
CBAND	2		CBAND(1-2) = IF FILTER BANDWIDTH (UPLINK, DOWNLINK) (DEFAULT UNIT IS HZ)			1.0E-8		SEC	
PBAND	2		PBAND(1-2) = PLL BANDWIDTH (UPLINK, DOWNLINK) (DEFAULT UNIT IS HZ)			125.		MHZ	
CBEAM	2		CBEAM(1-2) = TRANSMITTER BEAMWIDTH (UPLINK, DOWNLINK) (DEFAULT UNIT IS RADIANS)			125.		MHZ	
CSN	2		CSN(1-2) = SATCOM S/N THRESHOLD (UPLINK, DOWNLINK) (DEFAULT UNIT IS RATIO (DIMENSIONLESS))			1.5		DEC	
						15.		DB	
						15.		DB	

E. OPTICS CODE INPUTS

--TO RUN AN OPTICS PROBLEM THE USER MUST FIRST SET
OPTICS=REFER

--TWO TYPES OF OPTICS PROBLEMS CAN BE SIMULATED--
(1) A SURVEILLANCE PROBLEM WHERE THE SENSOR IS POINTED
AT SOME REFERENCE LOCATION (OTYPE=SURVEILLANCE), OR
(2) A BOOST TRACK PROBLEM WHERE A SENSOR LOCK IS
CREATED INTERNALLY AT A SPECIFIED TIME (OTYPE=TRACK).
--IN THE FIRST CASE, THE USER PROVIDES THE FIRST LOCK
TIME(CLOCK) AND THE REFERENCE POINT(REFPT) FOR
THE LOCK DIRECTION (OR TRACK THE BURST 1 FIREBALL)
(SEE SEASPT INPUT). IN THE SECOND CASE, THE USER INPUTS
THE BOOSTER MODEL AND THE TURNOUT POSITION, (BCPOS).

IN EITHER CASE, THE USER MUST SET UP AN OPTICAL SENSOR

1. RUN CONTROL

--SUCCESSFUL OPTICS LOCKS ARE CREATED INTERNALLY EVERY
(TIME) SECONDS.
--OPTICS OUTPUTS ARE CONTROLLED BY THE (OALC) PARAMETER
(1) FOR (OALC=POINTS), BOOST TRACK MEASUREMENTS ONLY
ARE PRODUCED, AND
(2) FOR (OALC=FOV), DATA STREAM OUTPUT AS THE DETECTOR
SCANS THE FOV WILL ALSO BE PRODUCED.
--THE BOOSTER MEASUREMENTS MAY BE USED TO INITIALIZE
(OR ACC TO) A TRACK FILE BY SETTING (TFILL=REFER), AND
THOSE MEASUREMENTS MAY BE FITTED WITH RADAR DATA BY
SETTING (SNLT=YES).

OPTICS 1 OPTICS = FLAG FOR INITIALIZING OPTICS CALCULATION (SET OPTICS=REFER 7EROS NO

DIRECTORY OF INPUT VARIABLES
Table A.1 (Continued)

INPUT VARIABLE	NO.	VALUES	DESCRIPTION	DEFAULT VALUES	UNIT-NAME ALLOWED
-------------------	-----	--------	-------------	-------------------	----------------------

OTYPE	1		FOR OPTICS CALCULATIONS)	TRACK	NO
OLOOK	1		OTYPE = OPTICS LOOK TYPE (TRACK OR SURVEILANCE)	99999.	YES
	1		CLOCK = TIME OF FIRST OPTICS LOOK (FOR OPTICS SURVEILANCE)		
FTIME	1		IS SEC)	10.	YES
CCALC	1		FTIME = FRAME TIME FOR OPTICS LOOKS (DEFAULT UNIT IS SEC)		
TFILE	1		CCALC = OPTICS CALCULATION TYPE (POINTS OR FOV)	FOV	NO
	1		TFILE = OPTICAL TRACK FILE FLAG (INPUT=REFER, FOR TRACK FILE/ --ZEROS FOR NO TRACK FILE)	ZEROS	NO
SNST	1		SNST = SENSOR SENSITIVITY FLAG (YES OR NO)	NO	NO
SENSPT	1		SENSPT = TYPE OF TARGET SENSOR IS POINTED TOWARD. USL REF FOR A FIXED POINT (HEFT). FIREBALL TO TRACK THE FIREBALL OF BURST. NOTE-- THE USL MUST ALSO SET DBREFER=NEFER AND CBTAG=REF-OR-JECT-- SEE SECTION C-3 ABOVE.	REF	NO
REPPT	4		REPPT(1-4) = REFERENCE POINT FOR SENSOR POSITIONING (HEFT(1-3)=POSITION COORD, REF(4)=COORD. TYPE -- GEOR, LOCKY?, OR RADAR) (DISTANCES IN MM, ANGLES IN DEG)	0. -79.33 47.75 GEOR	NO NO NO NO

2. SENSOR DATA

--THE SENSOR LOCATION CAN BE INPUT IN GEOGRAPHICAL
COORDINATES (GEOR) OR RELATIVE TO THE REFERENCE
LOCATION IN SECTION A. ABOVE.
--THERE ARE TWO WAVELENGTH BANDS ALLOWED AND THEY
BUILT IN SENSOR PROCESSING MODELS. THE FIRST TWO MODELS
SHOULD BE USED IN SURVEILANCE APPLICATIONS AND PROVIDE
SLIGHTLY DIFFERENT OUTPUT. THE THIRD MODEL (SURVEIL-04)
PROVIDES TRACK MEASUREMENT OUTPUT AND SHOULD BE USED
WHEN (TFIL=HEFT).

SNPOS	4		SNPOS(1-4) = OPTICAL SENSOR POSITION (SNPOS(1-3)=POSITION, COORD. SNPOS(4)=COORD. TYPE--GEOR, LOCKY?, OR RADAR) (DISTANCES IN MM, ANGLES IN DEG)	3787. -79.33 47.75 GEOR	NO NO NO NO
WLO	2		WLO(1-2) = LOW END OF SENSOR WAVELENGTH BAND --(TWO BANDS ALLOWED) (DEFAULT UNIT IS CM)	2.5L-6	YES
WHI	2		WHI(1-2) = HIGH END OF SENSOR WAVELENGTH BAND (DEFAULT UNIT IS CM)	2.6L-6	YES
OFERR	2		OFERR(1-2) = FIXED PORTION OF OPTICS RANDOM MEASUREMENT ERRORS IN AE COORD (DEFAULT UNIT IS RADIANS)	2.6L-6	YES
OSNR	2		OSNR(1-2) = S/N DEPENDENT PORTION OF OPTICS RANDOM MEASUREMENT ERRORS IN AE COORD (DEFAULT UNIT IS RADIANS)	2.7E-6	YES
OPUCL	1		OPUCL = OPTICAL SENSOR PROCESSING MODEL --(SURVEIL-01, SURVEIL-02, OR SURVEIL-04)	.01 .01 1. SURVEIL-01	MRAD MRAD MRAD NO

3. BOOSTER DATA

--TWO BOOSTER STAGES ARE ALLOWED. NOTE THAT THE TIME
COORDSP. TO THE INITIAL BOOSTER STATE IS SET INTERNALLY
TO 0. SEC AND THE RV IMPACT TIME SPECIFIED IN THE

DIRECTORY OF INPLT VARIABLES
 Table A.1 (Continued)
 INPUT NO.
 VARIABLE VALUES
 DESCRIPTION
 02/04/60 10.36.18. PAGE 10
 UNIT-NAVF
 ALLOWED

TRAJECTORY INPUTS (SEE SECTION C.3) WILL BE ADJUSTED
 ACCORDINGLY.

VARIABLE	INPUT NO.	VALUES	DESCRIPTION	UNIT	ALLOWED
FUEL	2	FULL(1-2) = FUEL TYPE (LIQUID OR SOLID)--NOTE--TWO STAGES ALLOWED		LIQUID	NO
THRST	2	THRST(1-2) = BOOSTER STAGE THRUST (DEFAULT UNIT IS GM)		1100000. LB	YES
WTI	2	WTI(1-2) = INITIAL STAGE WEIGHT (DEFAULT UNIT IS GM)		135000. LB	YES
WTF	2	WTF(1-2) = FINAL STAGE WEIGHT (DEFAULT UNIT IS GM)		70000. LB	YES
ANQZ	2	ANQZ(1-2) = NOZZLE EXIT AREA (DEFAULT UNIT IS CMSQ)		20000. LB	YES
TSUMN	2	TSUMN(1-2) = STAGE BURN TIME (DEFAULT UNIT IS SEC)		35000. LB	YES
REFR	2	REFR(1-2) = REFERENCE AREA FOR AERODYNAMIC DRAG CALCULATION (DEFAULT UNIT IS CMSQ)		8000. LB	YES
CRM0	2	CRM0(1-2) = AXIAL FORCE COEFFICIENT AT M=0.5		3000. INSC	YES
CRM1	2	CRM1(1-2) = AXIAL FORCE COEFFICIENT FOR M=1.0		0. SEC	YES
CRM3	2	CRM3(1-2) = AXIAL FORCE COEFFICIENT FOR M=3.0		0. SEC	YES
				35. FTSC	YES
				.10	NO
				.03	NO
				.19	NO
				.136	NO
				.11	NO
				.068	NO

DESCRIPTION OF USER INPUT AND COMMAND FORMATS . . .

Table A.1 (Concluded)

THE BASIC FORM FOR EACH INPUT LINE IS . . .

I1,I2,I3,.....

(ALL BLANKS ON THE LINE ARE IGNORED)
WHERE THE I1, I2, ETC., ARE EITHER COMMANDS OR ITEMS OF THE FORM . . .

ITEM=LIST

WHERE ITEM IS ONE OF THE INPUT VARIABLES OR VECTOR ELEMENTS AND LIST IS A LIST OF ONE OR MORE VALUES TO BE INPUT, STARTING AT THE NAMED ELEMENT. THE VALUES NEED NOT INCLUDE DECIMAL POINTS FOR WHOLE NUMBERS AND MAY BE APPENDED WITH APPROPRIATE UNIT NAMES IF ALLOWED FOR THAT VARIABLE. VALUES ARE SEPARATED BY COMMAS.

THE RECOGNIZED COMMANDS ARE . . .

ABORT CAUSES PROGRAM ABORT WITH NO OUTPUT FILE
(TO AVOID SUBMITTING A BATCH JOB)

CHANGELISTON TURNS ON SUBSTITUTION LIST OPTION (SHOWS HOW
VALUES ARE USED IN ROSCOE INPUT DECK)

CHANGELISTOFF TURNS CHANGELISTON OPTION OFF

HELP TO PRODUCE THIS MENU AGAIN

RUN TERMINATES EXECUTION AND PRODUCES OUTPUT FILE
FOR ROSCOE EXECUTION. ALTERNATE FORMS ARE
ENC OR END DATA

TABLE A.2

ALLOWABLE UNIT NAMES

<u>Category</u>	<u>Unit Name</u>	<u>Scaling Factor to Internal (Default) Units</u>
Frequency	MHZ	1,000,000
	KHZ	1,000
Time	HRS	1 (This may only be used for time-of-day inputs)
	SEC	1
Mass	KG	1,000
	GM	1
	LB	453.592
Ballistic Coeff.	PSF	0.4882405
	GM/CMSQ	1
Length	CM	1
	FT	30.48
	KM	100,000
	NMI or NM	185,325
	M	100
	KFT	30,480
Acceleration	G	980.665
Area	CMSQ	1
	MSQ	10,000
	INSQ	6.4516
	FTSQ	929.0304

TABLE A.2 (Cont'd.)

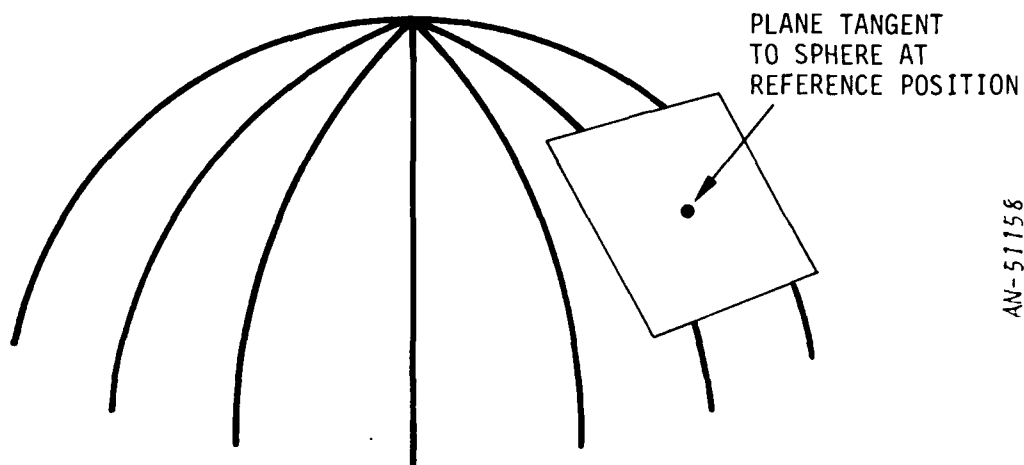
ALLOWABLE UNIT NAMES

<u>Category</u>	<u>Unit Name</u>	<u>Scaling Factor to Internal (Default) Units</u>
Yield	MT	1
	KT	0.001
Radar Range/Standard Target	CMSQCM	1
	KMSQM	10,000
	NMSQM	18532.5
	KFSQM	3048
Power	WATTS	10,000,000
Power Ratio	DB	$X \text{ dB} \rightarrow 10^{X/10}$
Angle	DEG	0.01745329252
	RAD	1
	MRAD	0.001

TABLE A.3

POSITION COORDINATE SPECIFICATIONS

GEOGR	<p>Geographical Coordinates:</p> <ul style="list-style-type: none">• Altitude (KM)• East longitude (DEG) (longitudes west of Greenwich input as negative)• North latitude (DEG) (south latitudes negative)
LOCXYZ	<p>Local Tangent Plane Coordinates (see Fig. A.1):</p> <ul style="list-style-type: none">• Geographic east (KM) (west input as negative)• Geographic north (KM) (south input as negative)• Distance above plane (KM)
RADAR	<p>Local Radar Coordinates (see Fig. A.1);</p> <ul style="list-style-type: none">• Slant range (KM)• Azimuth (DEG) (positive CCW from east)• Elevation (DEG) (positive above horizontal)



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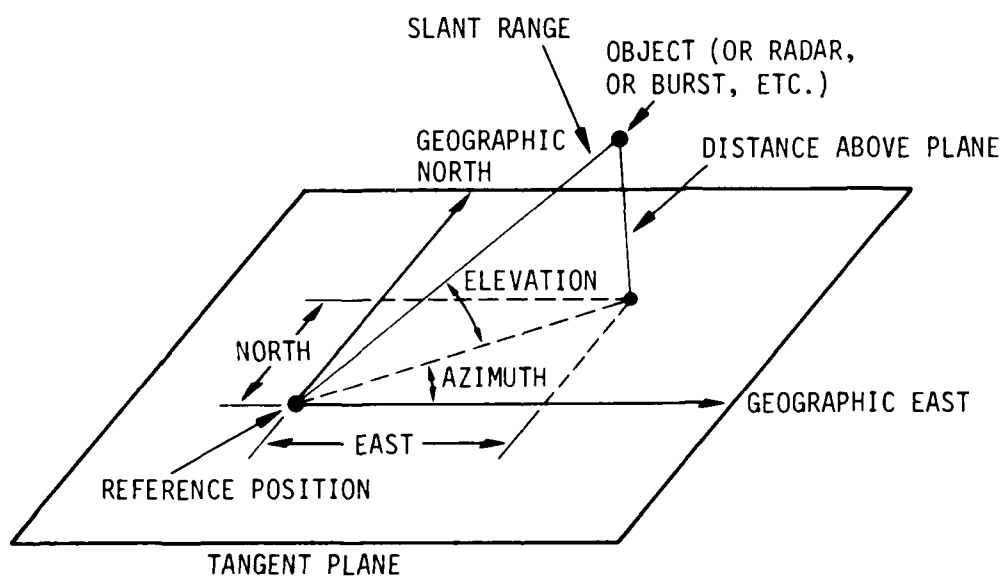


Figure A.1 Definition of Coordinates for Relative Coordinate Systems

TABLE A.4

SAMPLE CONTROL CARD DECK FOR AFWL/NOS/BE1

```

JOB CARD . . . . .
ACCOUNT CARD . . . . .
MAP(OFF)
ATTACH(XX1,OBINARY, ID=GRCXJJB, CY=1)
COPYBK(XX1,OBIN,240)
ATTACH(XX2,OBINARY, ID=GRCXJJB, CY=2)
COPYBF(XX2,OBIN)
RETURN(XX1,XX2)
REWIND(OBIN)
ATTACH(BCPYL,BCPYLRSCOE, ID=GRCXJJB, CY=3)
ATTACH(STRUCT,OSTRUCT, ID=GRCXJJB)
UPDATE(P=STRUCT,F,D,8,C=TAPE1,L=1)
BCPYL(TAPE1,OBIN,LFILE,,,READ1,REWIND,ERRORS)
RETURN(TAPE1,TAPE4,BCPYL,OBIN)
ATTACH(SPINE,SPINERSCOE, ID=GRCXJJB, CY=2, PR=1)
ATTACH(LTAG,LTAGRSCOE, ID=GRCXJJB, CY=5)
SPINL(,,,TAG,DATA,,,FATC,REWIND)
ATTACH(RLIBE,RLIBERSCOE, ID=GRCXJJB)
RETURN(TAPE1,TAPE2,TAPE3,TAPE4,TAPE5,TAPE6)
ATTACH(AMALGMB,AMALGMBRSCOE, ID=GRCXJJB)
AMALGMB.
RETURN(AMALGMB)
LOSET(LIL=RLIBE,PRESET=ZERO,FILES=TAPE1)
LOAD(LFILE)
NOGO.
RETURN(LFILE)
RETURN(RLIBE)
ATTACH(TAPE3,NEWDATERSCOE, ID=GRCXJJB)
SENER.
7-8-9 CARD
*IDENT DCHG
*COMPILE STRUCT
ANY MODS TO OSTRUCT FILE GO HERE . . . . .
7-8-9 CARD
CHANGELIST ON . . . . . (OPTIONAL)
SPINL DATA INPUTS . . . . .
. . . . .
. . . . .
RUN
7-8-9 CARD
6-7-8-9 CARD

```

TABLE A.5

SAMPLE PROCEDURE PERMFILE FOR INTERACTIVE USE

```

.PROC,ROSCUTS
COPYCR(ROSCUTS,DATEIR,2)
ATTACH(SPINE,SPINEROSCOE,ID=GRCXJJB,CY=2,NR=1)
ATTACH(INTAB,DATEIRROSCOE,ID=GRCXJJB,CY=5)
SPINL.
RETURN(SPINE,INTAB,NAFILL)
ZAP(DATEIR,WWW,IN)
COMMENT. FILE HAS BEEN BATCHED TO INPUT.
7-8-9 CARD
JOB CARD . . . . .
ACCOUNT CARD . . . . .
MAP(OFF)
ATTACH(XX1,OBINARY,ID=GRCXJJB,CY=1)
COPYBR(XX1,OBIN,240)
ATTACH(XX2,OBINARY,ID=GRCXJJB,CY=2)
COPYBR(XX2,OBIN)
RETURN(XX1,XX2)
REWIND(OBIN)
ATTACH(BCPYL,BCPYLROSCOE,ID=GRCXJJB,CY=3)
ATTACH(STRUCT,OSTRUCT,ID=GRCXJJB)
UPDATE(PE=STRUCT,F=D,B,C=TAPE1,L=1)
BCPYL(TAPE1,OBIN,LFILE,,,READ1,REWIND,ERRORS)
RETURN(TAPE1,TAPE4,BCPYL,OBIN)
COPYCR(INPUT,INDATA)
REWIND(INDATA)
ATTACH(RLIBE,RLIBEROSCOE,ID=GRCXJJB)
RETURN(TAPE1,TAPE2,TAPE3,TAPE4,TAPE5,TAPE6)
ATTACH(AMALGMB,AMALGMBROSCOE,ID=GRCXJJB)
AMALGMB.
RETURN(AMALGMB)
LOSET(LIB=RLIBE,PRESET=ZERO,FILES=TAPE1)
LOAD(LFILE)
NOGO.
RETURN(LFILE)
RETURN(RLIBE)
ATTACH(TAPE3,NEWDATEROSCOE,ID=GRCXJJB)
SENER.
7-8-9 CARD
*IDENT OCHG
*COMPILE STRUCT
ANY ADDS TO OSTRUCT FILL GO HERE . . . . .
7-8-9 CARD
6-7-8-9 CARD

```

TABLE A.6
TIME-SHARE INPUTS
(Underlined portions typed by User)

1.	COMMAND	-	<u>ATTACH (ROSCOTS, ID = GRCXJJB)</u>
2.	COMMAND	-	<u>RØSCØTS</u>
3.	INPUTS?		(USER TYPES IN INPUTS)
	INPUTS?		(USER TYPES IN INPUTS)
	.		.
	.		.
	.		.
	[ERRORS	- (--IF THERE ARE INPUT ERRORS, RØSCØTS LISTS THEM HERE AND REQUESTS INPUTS AGAIN)
	INPUTS?		<u>RUN</u>

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